

Advanced Post Processing: How to Efficiently Share Your FEA Results

A Simcenter Femap Seminar for Simulation Engineers

Geoffrey Bee, PE - Application Engineer, CAE Adrian Jensen, PE, MBA - Director of Engineering, CAE





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October 31st – November 10th, 2022 Live, Online <u>AppliedCAx.com/Training</u>

CAE Support Review:

As far as tech support is concerned, I have had fast and top-quality responses. The awesome thing is, I get a lot of information during the support communication, but I also receive the full concept and learn a lot. Even if the issue is very simple, I get a quick response. If someone asks me about buying Siemens products, I will surely recommend Applied CAx.

Srivatsa Pradeep, MSME Project Consultant (Structures & FEA) Hatch LTK Engineering Services

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Simcenter Femap Practices: Advanced Post Processing in Femap v2022.2

A Seminar for Simulation Engineers Geoffrey Bee, PE – Application Engineer, CAE Adrian Jensen, PE, MBA – Director of Engineer, CAE



Seminar Outline



Output Management

- Scratch and Output File Organization
- Nastran Output Requests
- File Options
 - Attach to Results
 - Create (Analysis) Studies
- Things to look for in the f06 file
 - OLOAD
 - SPCForce

PostProcessing Toolbox

- Deform View Style
 - Deformed
 - Animate
 - Vector
- Contour View Style
 - Contour
 - Beam Diagram
 - Vector
 - Freebody View Style
 - Freebody
 - Interface Load
 - Section Cut

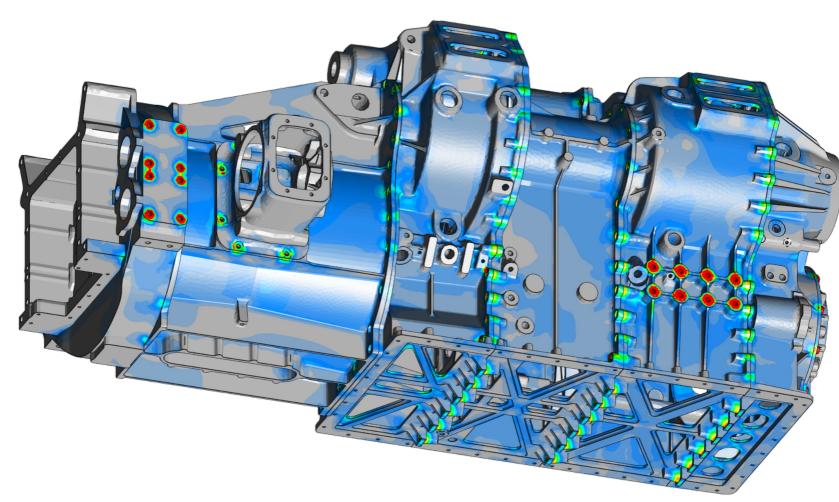
Charting

•

- Expanding Complex Results
- Charting Complex Results

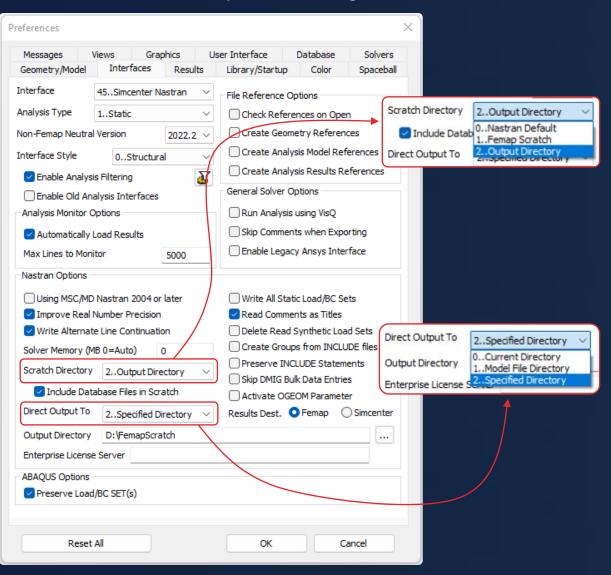
Femap API

- Group_Post_FNO
- ASME Stress Intensity Calculator





Scratch and Output File Organization



Scratch Directory

O..Nastran Default: Directory chosen during installation to use for creating Simcenter Nastran scratch files.

1..Femap Scratch: Directory specified in the Database tab of the Preferences dialog box, where the FEMAP has been directed to place the FEMAP scratch file.

2..Output Directory: Directory specified by the Direct Output To option on this tab of the Preferences dialog box.

Direct Output To

O..Current Directory: Last used directory by FEMAP. If a model has been saved to a directory, the output will be directed to that directory when this option is on.

1..Model File Directory (default): The directory where the model file is located. All output will go into this directory until the model is saved somewhere else.

2..Specified Directory: This option allows you to send all Simcenter Nastran output to a directory that you have specified. This can helpful because your output will always be in the same place if you need to view the files or "clean-up" leftover output files from old analysis runs.



Femap Scratch Files

Geometry/Model		faces	Results	Library/Startup		Spaceball
Messages	Views	Grap	ohics	User Interface	Database	Solvers
	Data	oase Pre	ferences ar	e only applied at sta	artup.	
Database Optio	ns			Timed Save		
Backup befo	re Save (in	nmediate)	On	Interval	10
Cleanup Duri			e)	Notify	Commands	25
Preserve Ne	xt ID durin	g Rebuik	ł	Windows Explor	er Data	
< Reset Next I	D after De	lete All		Save Thun	nbnail	
Low Disk Warnir	ng (0=Nev	er)	10 MB	Save Size	and Notes Info)
Undo Levels	2	0		Scratch Director	ry	
Database Perfo		12944	MByte	scratch file directory s	tch directory is s will be placed pecified by the nment variable tch	in the TEMP
				D. V emapsera		
Max Cached La	bel 99	9999999		Recover	Scratch Directo	ry
Blocks/Page	4			Recove	er _DBData File	
Open/Save Me	thod 0	Windows	sI/O ∨			
	R	ead/Writ	e Test	Databa	se Options Help	o

- 5	ystem variables					
	Variable	Value				
	PROCESSOR_REVISION	8d01				
	PSModulePath	%ProgramFiles%\WindowsPowe	erShell\Modules;0	C:\Windows\syste	m32\Wind	
	SPLM_LICENSE_SERVER	28000@LOCALHOST				
	TEMP	C:\Windows\TEMP				
	TMP	C:\Windows\TEMP			1	
	UGII_BASE_DIR	C:\Program Files\Siemens\NX20	07			
	UGII_LANG	english				
	USERNAME	SYSTEM				
	windir	C:\Windows				
	ZES_ENABLE_SYSMAN	1				
			New	Edit	Delete	

Reset All

Cancel

OK



Nastran Output Requests

NASTRAN Output Rec	quests			×
Nodal				
Displacement	0Full Model	~	Velocity	0Full Model 🗸 🗸
Applied Load	0Full Model	\sim	Acceleration	0Full Model 🗸 🗸
Constraint Force	0Full Model	~	Kinetic Energy	0Full Model 🗸 🗸
Equation Force	0Full Model	\sim	Temperature	0Full Model 🗸 🗸
Force Balance	0Full Model	\sim		
Elemental				
Force	0Full Model	\sim	Heat Flux	0Full Model 🗸 🗸
□ Stress	0Full Model	\sim	Enthalpy	0Full Model 🗸 🗸
🗌 Total Strain	0Full Model	\sim	Enthalpy Rate	0Full Model 🗸 🗸
Elastic Strain	0Full Model	\sim	Temperature	0Full Model 🗸 🗸
Thermal Strain	0Full Model	\sim	Kinetic Energy	0Full Model 🗸 🗸
O Fiber	○ Curvature		Energy Loss	0Full Model 🗸 🗸
🕑 Strain Energy	0Full Model	~	🗌 Fluid Pressure	0Full Model 🗸
Contact				
Contact	0Full Model	~	Glue	0Full Model 🗸
Customization				
< Element Corner Res	sults	F	Results Destination	Prev
Output Modes (a,b,c 1	THRU d)		2PostProcess Only	<u> </u>
			0Default 1Print Only	ОК
O Magnitude/Phase	O Real/Imaginary	E	2PostProcess Only 3Print and PostProcess	Cancel
Relative Enforced N	1otion Results		4Punch Only 5Punch and PostProcess 6XDB	_

Customization also allows you to select a results destination file type:

Results Destination

- 0..Default = .op2 files created using PARAM,POST,-1
- 1..Print Only = *.f06 An ASCII output file you will use most frequently
- 2..PostProcess Only = *.op2 A binary file known as an "OUTPUT2" file
- 3..Print and PostProcess = *.f06 and *.op2
- 4..Punch Only = *pch An ASCII file that contains "punched" output
- 5..Punch and PostProcess = *.pch and *.op2
- 6..XDB = *.xdb A binary file known as a "results database" file

Note:

- When you select "3..Print and PostProcess" as the Results Destination, you are sending the results to both the .f06 and the .op2 file. Normally, you would not want to do this, but the option is there to complete all the possible combinations for requesting output.
- When FEMAP runs Simcenter Nastran, it automatically reads the results. This can be changed by: File, Preferences, then click Interfaces, then uncheck box "Automatically Load Results".
- FEMAP reads the .f06 file first to obtain any error, warning, or information messages that might have occurred during the analysis.



Nastran Output Requests

NASTRAN Output Requ	Jests					×
Nodal						
🗹 Displacement	0Full Model	\sim	Velocity		0Full Model	2
Applied Load	0Full Model	\sim	Acceleration		0Full Model	2
Constraint Force	0Full Model	\sim	Kinetic Energy		0Full Model	2
Equation Force	0Full Model	\sim	Temperature		0Full Model	2 /
Force Balance	0Full Model	~				
Elemental						
Force	0Full Model	\sim	Heat Flux		0Full Model	2
Stress	0Full Model	\sim	Enthalpy		0.,Full Model	2
Total Strain	0Full Model	~	Enthalpy Rate		0Full Model	2
Elastic Strain	0Full Model	\sim	Temperature		0Full Model	2
Thermal Strain	0Full Model	\sim	Kinetic Energy		0Full Model	2
O Fiber	○ Curvature		Energy Loss		0Full Model	2
Strain Energy	0Full Model	\sim	Fluid Pressure		0Full Model	2
Contact						
Contact	0Full Model	2	Glue		0Full Model	лI
Customization						
Element Corner Resu	ilts	Re	esults Destination		Prev	
-			2PostProcess Only	+		
Output Modes (a,b,c Th	чко и)		Femap Simo	ent	ег ОК	
Magnitude/Phase	O Real/Imaginary	_			Cancel	51
	· · · ·	Ec	ho Model			
Relative Enforced Mo	otion Results					

As of version 2022.1 it is now possible to instruct Nastran to create an output file, in particular an .op2 file, which can be used by other applications in the Siemens Software Portfolio, including Simcenter 3D.

Once selected, Nastran knows to produce this type of output by having PARAM, POST set to a value of -2 in the input file.



Nastran Output Requests

references	×
Messages Views Graphics	User Interface Database Solvers
Geometry/Model Interfaces Results	Library/Startup Color Spaceball
Interface 45Simcenter Nastran	File Reference Options
Analysis Type 6Random Response	Check References on Open
Non-Femap Neutral Version 2022.2	Create Geometry References
Interface Style 0Structural	Create Analysis Model References
🗹 Enable Analysis Filtering 🛛 🛛	Create Analysis Results References
Enable Old Analysis Interfaces	General Solver Options
Analysis Monitor Options	Run Analysis using VisQ
Automatically Load Results	Skip Comments when Exporting
Max Lines to Monitor 5000	Enable Legacy Ansys Interface
Nastran Options	
Using MSC/MD Nastran 2004 or later	Write All Static Load/BC Sets
Improve Real Number Precision	Read Comments as Titles
Write Alternate Line Continuation	Delete Read Synthetic Load Sets
Solver Memory (MB 0=Auto) 0	Create Groups from INCLUDE files
Scratch Directory 2Output Directory	Preserve INCLUDE Statements Skip DMIG Bulk Data Entries
🗹 Include Database Files in Scratch	
Direct Output To 2Specified Directory	Results Dest. O Femap O Simcenter
Output Directory D:\FemapScratch	
Enterprise License Server	
ABAQUS Options	
Preserve Load/BC SET(s)	
Reset All	OK Cancel

 You can also set Femap to always create .op2 files using PARAM,POST,-2 automatically.

To do this: File \rightarrow Preferences. In the Preferences dialog box, choose the Interfaces tab, and set the Results Destination option to Simcenter.

Some pre and post processors that can read .op2 files created using PARAM,POST,-1 (Femap default) require that the model information also be in the .op2 file, which is done by setting PARAM,OGEOM to a value of YES.

By turning on the Activate OGEOM Parameter option on this same tab, all newly created Analysis Sets will have the OGEOM option turned on automatically.



File Options - Attach to Results

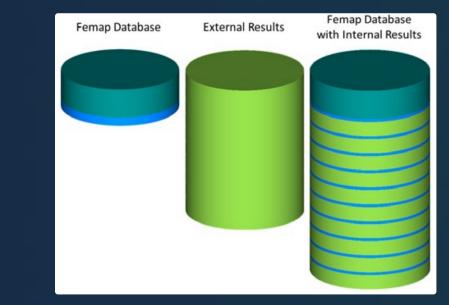
	Manage	Results Files		– 🗆 X
		1 Attached File(s)		Attach Options
	ID 1	Filename frame_beam_model_l	File Path D:\FemapScratch\	Attach Options File Format Simcenter Nastran MSC/MD Nastran Femap FNO/Autodesk Nastran Comma Separated Abaqus Memory Mapped File Attach File Unload
File	Info			Reload
Sol	ver: alysis Typ			Detach
	e / Label			Detach All
				Find File
File	Date:			Save To Model
Mer	mory Ma	pped:		Done

Note: Within File > Preferences you can choose to automatically attach to results files and use "Memory Mapping" for a potential increase in speed. When Memory Mapping is turned on, FEMAP will attempt to attach to the file using RAM. In order for an attached output file to be properly "Memory Mapped" it must be able to fit into a contiguous block of unused system memory. One can verify "Memory Mapped" status in the File Info section

File \rightarrow Attach to Results... attaches to results files and allows post processing to occur without "internalizing" the contents of each file into the FEMAP database. This is especially helpful when you have a large output file. In general, there are two reasons for large output files:

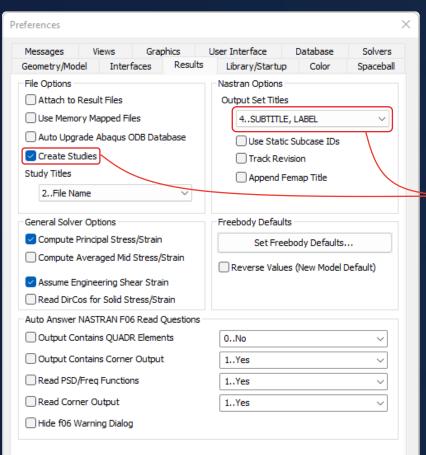
- First, a large model will typically create a large output file unless care has been taking in requesting output.
- Second, analysis types which create a large number of Output Sets, such as transient response, frequency response, and nonlinear analysis, can create large output files for even small and medium sized models.

Maintaining external results data in this way minimizes the Femap database size and makes results data handing much more manageable.

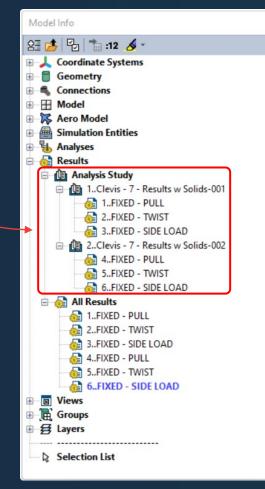




File Options - Create (Analysis) Studies



You can use Analysis Studies to group your output sets for better organization and data processing.



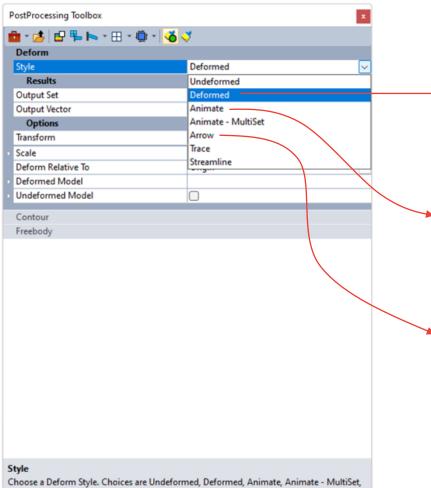
Reset All

Cancel



Deform Style

Arrow, Trace, and Streamline,



Showing the deformed shape of your model is a standard first step in post processing. It's a quick an easy way to verify the application of your boundary conditions and see how the structure deforms under load.

Deformed

This is the standard option. It provides a static, deformed image of the structure at the chosen scale factor. Be careful when choosing output vectors; you will almost always want to use "1..Total Translation".

Animate

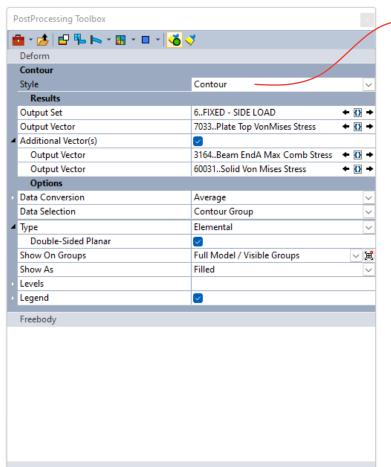
As the name implies, this options animates the deformation of the model. A single output set or, in the case of nonlinear analysis, multiple sets can be animated. The user can control the speed and resolution of the animated view.

Arrow

This option is useful for detailed investigation of displacements. The Arrow style will display displacement vectors on each node. The user can control the magnitude and view style of the arrows.

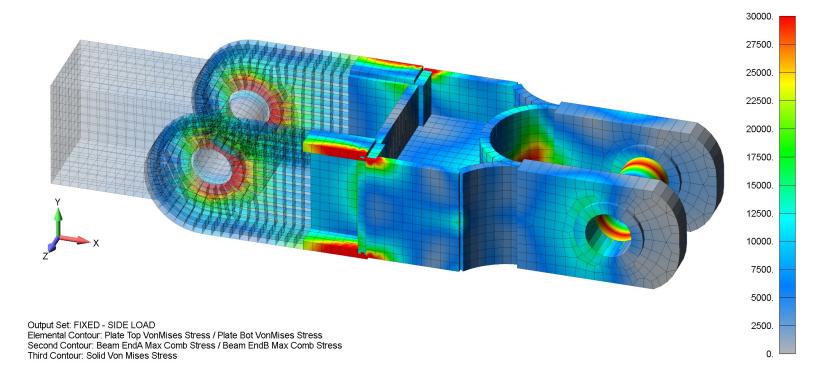


Contour Style - Contour



<u>Contour</u>

The Contour style allows the user to overlay displacement, force, strain, and stress information on the mesh



Style

Choose a Contour Style. Choices are No Contours, Contour, Criteria, Beam Diagram, IsoSurface, Section Cut, or Contour Arrow.



Contour Style – Beam Diagram

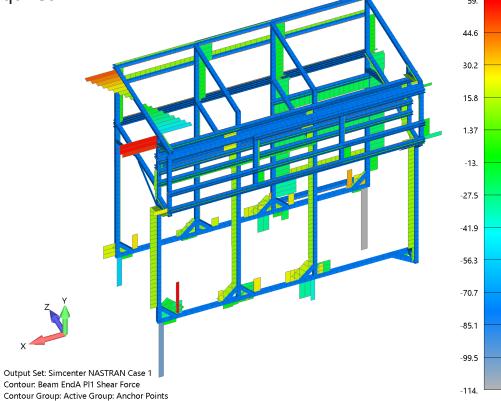
P	ostProcessing Toolbox		
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	Deform		
	Contour		
	Style	Beam Diagram	\sim
	Results		
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	Additional Vector(s)		
	Options		
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	Show On Groups	Active Group	Ħ
4	Show As	Beam Diagram	\sim
	Label	No Labels	\sim
	Direction	Element Y	\sim
	Show Reversed		
	Scale %	10	
	Border Color	62	
Þ	Levels		
Þ	Legend	2	
	Freebody		

Show As

Choose to display a Beam Diagram plot as a Beam Diagram or a Beam Contour.

Beam Diagram

Remember those shear-moment diagrams from all those years ago? The Beam Diagram view style keeps the tradition alive! Display forces, moments, displacements and stresses contoured over the mesh or plotted beam diagram style. Note, beam elements required.





Contour Style – Contour Arrow

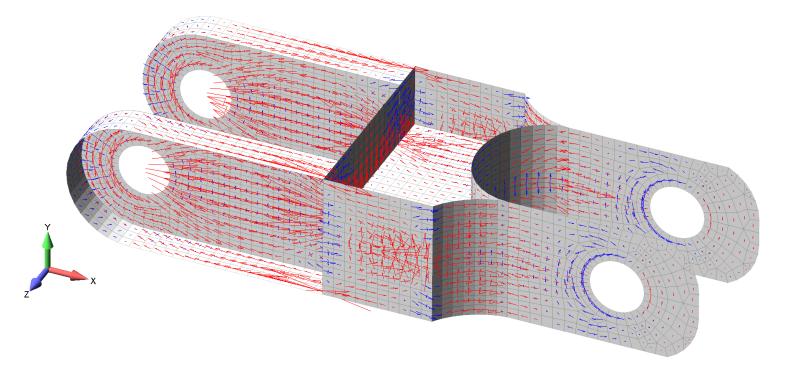
PostProcessing Toolbox		
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Deform		
Contour		
Style	Contour Arrow	~
Results		
Output Set	1FIXED - PULL 🗲 🚺	•
Output Vector	7027Plate Top MinorPrn Stress 🗲 🚺	+
Options		
Select Arrows from Contour Vector		
Arrows		
Arrow Type	2D Components	
X Arrow Display / Vector Select	7026Plate Top MajorPrn Stress	
Y Arrow Display / Vector Select	7027Plate Top MinorPrn Stress	
Transform		Z
Data Selection	Contour Group	~
Show On Groups	Full Model / Visible Groups	٣,
Arrow Head and Color	Auto	
Solid Arrows		
Arrow Length	0.01	
Min Vector Magnitude	✓ 1.E-8	
Arrow Labels	Off	~
Levels		
Legend		

Legend

On by default. When On, the Contour/Criteria Legend will be displayed. When Off, no Legend will be visible.

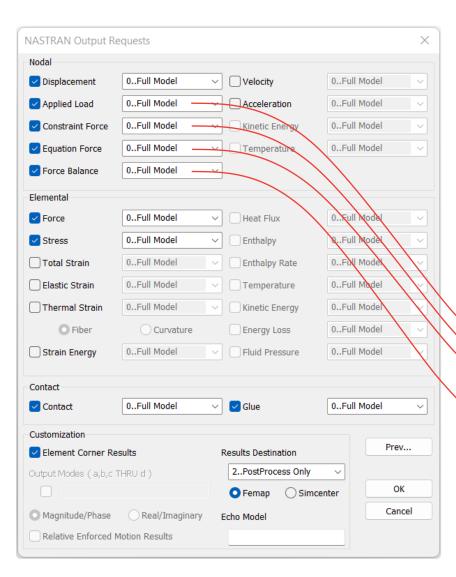
Contour Arrow

This view style is most useful for looking at the directional flow of stress with the principal stress output vectors. The Vector view style places vectors on the elements to show the direction and magnitude of stresses and forces.





Freebody View Style - New Freebody



Freebody Diagrams (or FBDs) are a somewhat advanced post technique, yet they are beautifully simple at the same time. It's all about summing the forces and moments for a selection of elements and nodes. The trick is carefully selecting the entities and making sure that the "Freebody Contributions" are logical.

Step one is ensuring you have requested the appropriate output for the FBD you intend to create prior to running the model

New Freebody			\times
ID 2 Title Display Mode	ce Load 🔿 Se	ction Cut	
Freebody Contributions	Vector Display Nodal Forces	1Display Components	~
 Reaction MultiPoint Reaction 	Nodal Moments	0Off	~
Peripheral Elements	Total Force Total Moment	1Display Components 0Off	~
Contact Glue Nodal Summation		s in Total Summation	Mz
Freebody None			
Default Settings	More	OK Cance	

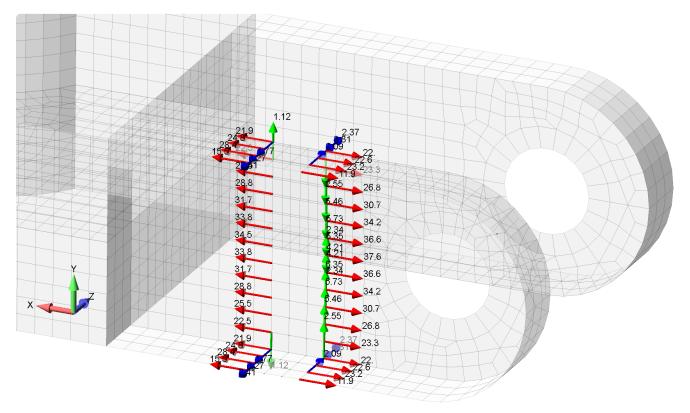


Freebody View Style - Freebody

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Contour	
Freebody	
Show Visible Freebodies	
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Reverse Freebody Values	
Freebody Properties	
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Is Visible	
Coordinate System	0Global Rectangular
Display Mode	 Freebody Interface Load Section Cut
Entities	
Entity Selection Mode	Entity Select 🗸
Freebody Elements	
Nodal Vector(s)	
 Freebody Contributions From 	
Applied	
Reaction	
MultiPoint Reaction	
Peripheral Elements	
Freebody Elements	
Contact	
Glue	
Nodal Summation	
Freebody Entity Colors View Properties	

Reverse Freebody Values

Reverses the value for all freebody forces and moments to show reaction instead of applied. Affects both displayed and listed data The default display mode for the FBD toolbox is simply called "Freebody". This display mode only requires the user to select elements. FBDs will be generated on any on the nodes of the selected elements that are connected to loads, constraints, RBEs, or other elements.

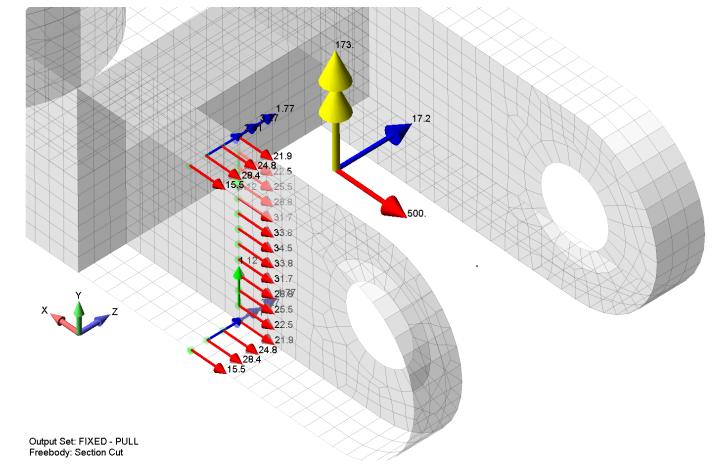




Freebody View Style - Interface Load

Pos	stProcessing Toolbox	
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F	reebody	
S	how Visible Freebodies	
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4	Freebody Properties	
4	Freebody	1Section Cut 🗸 👰
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	Coordinate System	0Global Rectangular 🛛 🗸
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	Display Mode	Interface Load
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	Entities	
	Entity Selection Mode	Entity Select 🗸
	Freebody Elements	
	Freebody Nodes	
	Total Summation Vector	₽ ₽ ₽ ₽ ₽ ₽ ₽ ₽ ₽ ₽
	Nodal Vector(s)	₽↑ ●↑ ●↑ ●
	Freebody Contributions From	
	Freebody Entity Colors	
	View Properties	

The "Interface Load" display mode will only generate FBDs on selected nodes attached to selected elements. In addition to the nodal FBDs, the Interface Load option allows the user to generate total summation vectors.



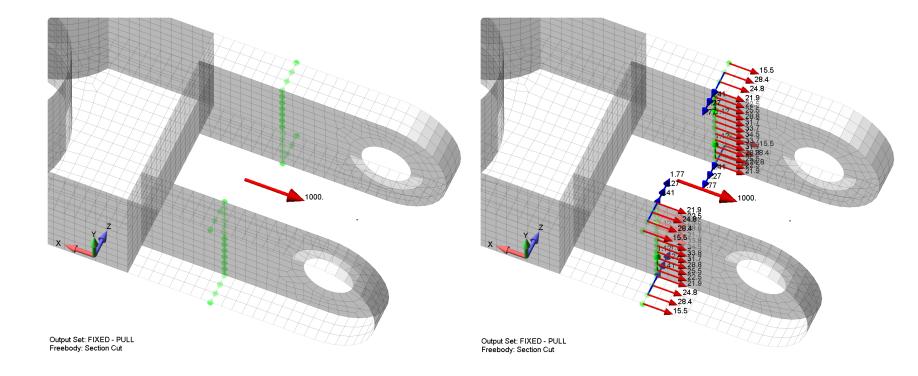


Freebody View Style – Section Cut

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Deform		
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Show Visible Freebodies		
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Freebody	1Section Cut	<u>~</u> 🛡
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ls Visible		
Coordinate System	0Global Rectangular	<u>~ </u>
	Freebody	
Display Mode	Interface Load	
	Section Cut	
 Entities 		_
Entity Selection Mode	Plane / Normal	<u> </u>
⊳ Plane	[0.887498,0.,0.] [1.,0.,0.]	
Location Slider	43	
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Plane Offset Percentage	10.	
> Total Summation Vector	₽ ₽ ₽ ₽ ₽ ₽ ₽ ₽	
Nodal Vector(s)		
Freebody Contributions From		
Freebody Entity Colors		

Entities

"Section Cut" display mode operates in the same manner as the Interface Load, but rather than selecting elements and nodes, the user simply selects a cutting plane. FEMAP will automatically select nodes along the cutting plane and elements on one side of the plane.



Charting



Charting – Expanding Complex Results

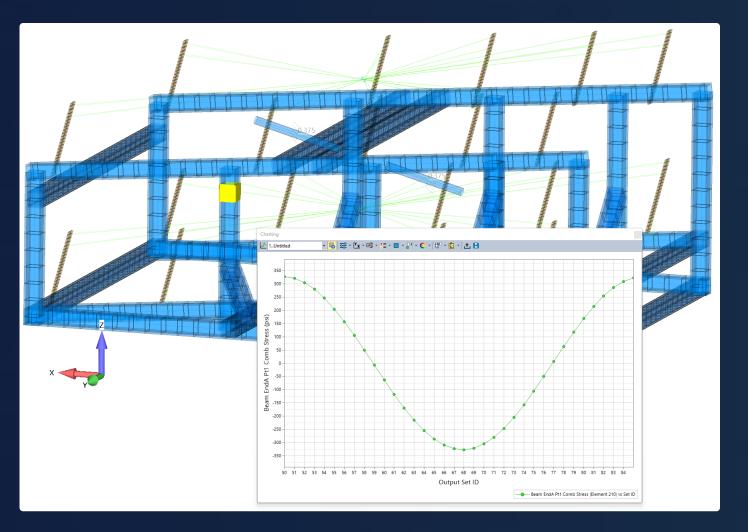
After running a Frequency/Harmonic Response analysis it is often useful to expand the complex results. Charting is a great way to visualize these expanded results.

📧 Analysis Set Manager (Active: 2Dynamic_Analysis) — 🗆 🗙 🗖 Simcenter Femap with Nastran - [Frequency.modfem] - [2021 Ap			Applied CA	Ax Femap View Settings]	Select Complex Output to Expand			– o x		
		File Tools Geometry Connect	Model Mesh Modi	ify List	Delete Group View Window H	Help	Model - Frequency.modfem			
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	Export	i 🔽 🕫 🕱 🛪 🖗 🖗 💿 🖸 🛛	Element Ctrl	I+E			1Mode 1, 5.025262 Hz			Quick Filter 0None - Ignore V
Analysis Set X	Active	Model Info	GU Material	21 /	Applied CAx Femap View Sett 🗙		🗌 🌆 3Mode 3, 22.37823 Hz		 2T1 Translation 3T2 Translation 	3154Beam EndB Pt4 Comb Stress 6043Plate Top Fiber
	Preview Input	2 🛃 🏄 🔂 👘 :12 🔏 🗸	· · ·				4Mode 4, 31.35041 Hz		4T3 Translation 6R1 Rotation	6044Plate Bottom Fiber 7020Plate Top X Normal Stress
Title Dynamic_Analysis		Coordinate Systems	Laminates	•			5Mode 5, 60.15744 Hz		7R2 Rotation	7021Plate Top Y Normal Stress
	MultiSet	🗊 🗐 Geometry	Load	- -			C.Mode 6, 67, 19438 Hz		8R3 Rotation	7023Plate Top XY Shear Stress
Analysis Program 36Simcenter Nastran V	Сору	🗈 🐔 Connections	Constraint				8Mode 8, 75.67046 Hz		22T1 Acceleration 23T2 Acceleration	7033Plate Top VonMises Stress 8006Plate X Membrane Force
Analysis Type 4Frequency/Harmonic Response V		🗈 🖽 Model	Analysis				9.,Mode 9, 76.38295 Hz		24T3 Acceleration	8007Plate Y Membrane Force
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OIntegrated Solver License Type 0Desktop ∨		🖨 🔠 Analyses	Simulation Entities	s 🕨			12Case 1 Freq 5.25		28R3 Angular Acceleration 52T1 Constraint Force	8012Plate Y Bending Moment 8013Plate XY Bending Moment
O Linked Solver	Load	3. Modal_Analysis	Optimization				13Case 1 Freq 5.5		53T2 Constraint Force	8014Plate X TransShear Force
Solver is undefined. Go to File Preferences Solvers.		2Dynamic_Analysis	Optimization	•					54T3 Constraint Force	8015Plate Y TransShear Force
	Save	🖻 🙀 Results	5xy Function						56R1 Constraint Moment	9020Plate Bot X Normal Stress 9021Plate Bot Y Normal Stress
⊖ VisQ		🕀 🕕 Analysis Study	Output) Ini	Create/Manage Analysis Study				58R3 Constraint Moment	9023. Plate Bot XY Shear Stress
		🖃 🚮 All Results	output		· · · · ·				250Complex Mode Shape	9033Plate Bot VonMises Stress
Next OK Cancel	New	1Mode 1, 5.025262 Hz		0	Create/Manage Set				3014Beam EndA Plane 1 Moment	10007PltC1 Top Fiber
		2Mode 2, 5.048708 Hz			Vector				3015Beam EndA Plane2 Moment 3016Beam EndB Plane1 Moment	10008PltC1 Bottom Fiber 11020PltC1 Top X Normal Stress
	Edit	3Mode 3, 22.37823 Hz			Define				3017Beam EndB Plane2 Moment	11021PltC1 Top Y Normal Stress
	Done	4Mode 4, 31.35041 Hz							3018Beam EndA Pl1 Shear Force	11023PltC1 Top XY Shear Stress
		5Mode 5, 60.15744 Hz			Fill				3019Beam EndA Pl2 Shear Force	11033PltC1 Top VonMises Stress
		6Mode 6, 67.19438 Hz		- 6	Process				3020Beam EndB Pl1 Shear Force 3021Beam EndB Pl2 Shear Force	12006PltC1 X Membrane Force 12007PltC1 Y Membrane Force
		7Mode 7, 75.0835 Hz		· · · ·	Calculate				3022Beam EndA Axial Force	12008PltC1 XY Membrane Force
					Calculate				3023Beam EndB Axial Force	12011PltC1 X Bending Moment
		8Mode 8, 75.67046 Hz		0	From Load				3024Beam EndA Torque	12012PtC1 Y Bending Moment
		9Mode 9, 76.38295 Hz			Transform				3025Beam EndB Torque 3026Beam EndA Warping Torque	12013PltC1 XY Bending Moment 12014PltC1 X TransShear Force
		10Mode 10, 84.05964 Hz			Extrapolate				3027Beam EndB Warping Torque	12015PltC1 Y TransShear Force
		🔄 11Case 1 Freq 5.							3139Beam EndA Pt1 Comb Stress	13020PltC1 Bot X Normal Stress
				() ()	Global Ply				3140Beam EndA Pt2 Comb Stress	13021PltC1 Bot Y Normal Stress
		🔂 13Case 1 Freq 5.5			Convert Complex				3141Beam EndA Pt3 Comb Stress 3142Beam EndA Pt4 Comb Stress	13023PltC1 Bot XY Shear Stress 13033PltC1 Bot VonMises Stress
		E Views			Expand Complex				3151Beam EndB Pt1 Comb Stress	14007PltC2 Top Fiber
		🕀 🕞 🔂 Groups			,				3152Beam EndB Pt2 Comb Stress	14008PltC2 Bottom Fiber
		⊕ ∰ Layers			Forced Response				3153Beam EndB Pt3 Comb Stress	15020PltC2 Top X Normal Stress
				_						
		Selection List								,
							Add Similar Layer/Ply Results	Add Compon	nent/Corner Results	OK Cancel

Charting



Charting – Charting Complex Results



The Charting Pane provides a simple interface for creating plots of output data as well as function data without relying on an external program, such as Excel.

It's a quick and easy way to get data into Excel as well if that's ultimately where you want the information.

Here, we can see the P1 Comb Stress in the square tube (highlighted) for one full cycle, or complete load reversal.

Femap API



Group_Post_FNO

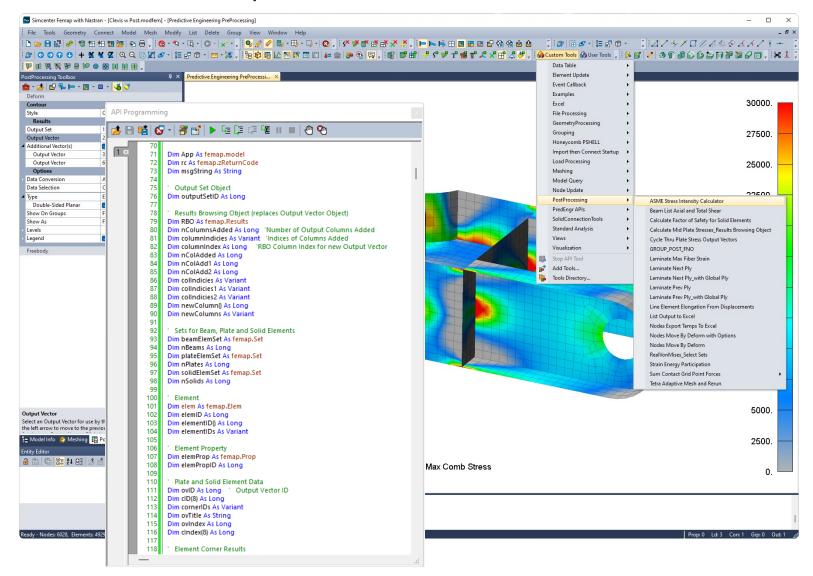
"Group_Post_FNO" is the API for you if your model generates extremely large output files. This tool can be used to export a portion of the results from an analysis. Working with smaller results subsets will improve performance and may be the only practical way of sending/sharing results via the internet if the output file is enormous.

	💽 Select Output to Export — 🗆 🗙
Export Options X	Model - Clevis Starting Point.modfem Output Sets Output Vectors
Select Export Method C Export Full Model K Cancel	Output Sets Output Vectors Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Construct of the sets Image: Consets Image: Consets <
Export Options X Select Group to Process - O X	3164Beam EndA Max Comb Stress 9026Plate Bot MajorPrn Stress 3165Beam EndA Min Comb Stress 9027Plate Bot MinorPrn Stress 3166Beam EndB Max Comb Stress 9029Plate Bot PrnStress Angle
Select Export Method IMid-Surface of I Export Full Model IMid-Surface of I Export Selected Group IMid-Surface of I	3167Beam EndB Min Comb Stress 9030Plate Bot Mean Stress 3168Beam Tension M.S. 9031Plate Bot MaxShear Stress 3169Beam Compression M.S. 9033Plate Bot VonMises Stress 6043Plate Top Fiber 60010Solid X Normal Stress 6044Plate Bottom Fiber 60011Solid Y Normal Stress 7020Plate Top X Normal Stress 60012Solid Z Normal Stress 7021Plate Top Y Normal Stress 60013Solid XY Shear Stress 7023Plate Top XY Shear Stress 60014Solid YZ Shear Stress
OK Cancel	Add Similar Layer/Ply Results Add Component/Corner Results OK Cancel

Femap API



ASME Stress Intensity Calculator



"ASME Stress Intensity Calculator" can also be found in the Custom Tools. This API calculates membrane stresses and stress intensity (think Tresca, not fracture mechanics) for plate elements. These output vectors are commonly used in the pressure vessel industry.

Additionally, if you need to create your own custom output vectors, this API could be a good starting point.